

Abstract to be submitted to The International
Conference On
Metallurgical Coatings and Thin Films 2004
Sponsored by the
Advanced Surface Engineering Division of AVS

April 19 - April 23, 2004
San Diego, California, USA

Development and Testing of Ceramic Thermal Barrier Coatings

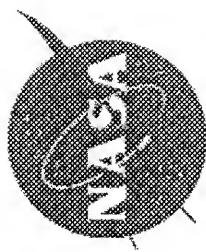
Dongming Zhu, Sung R. Choi, Robert A. Miller
NASA John H. Glenn Research Center
21000 Brookpark Road
Cleveland, OH 44135

Ceramic thermal barrier coatings will play an increasingly important role in future gas turbine engines because of their ability to effectively protect the engine components and further raise engine temperatures. Durability of the coating systems remains a critical issue with the ever-increasing temperature requirements. Thermal conductivity increase and coating degradation due to sintering and phase changes are known to be detrimental to coating performance. There is a need to characterize the coating behavior and temperature limits, in order to potentially take full advantage of the current coating capability, and also accurately assess the benefit gained from advanced coating development. In this study, thermal conductivity behavior and cyclic durability of plasma-sprayed ZrO_2 -8wt% Y_2O_3 thermal barrier coatings were evaluated under laser heat-flux simulated high temperature, large thermal gradient and thermal cycling conditions. The coating degradation and failure processes were assessed by real-time monitoring of the coating thermal conductivity under the test conditions. The ceramic coating crack propagation driving forces and resulting failure modes will be discussed in light of high temperature mechanical fatigue and fracture testing results.

Full paper version
and advanced copy of this Poster
already approved by Diane Chapman

Development and Thermal Fatigue Testing of Ceramic Thermal Barrier Coatings

Dongming Zhu, Sung R. Choi and Robert A. Miller

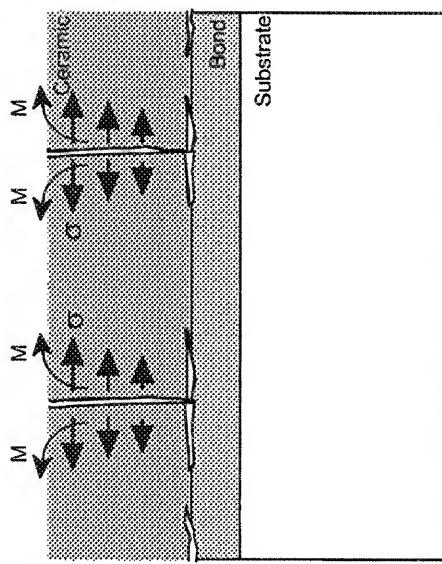


**NASA John H. Glenn Research Center
21000 Brookpark Road, Cleveland, OH 44135**

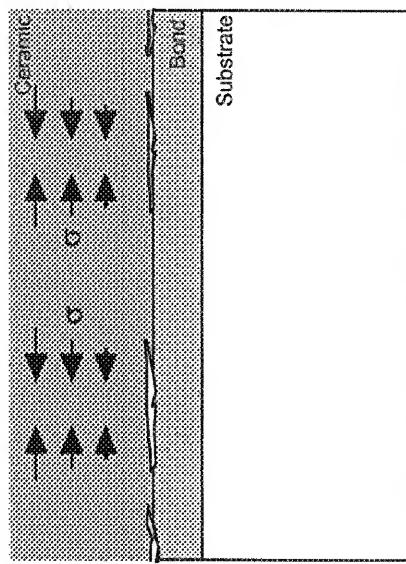
The International Conference On Metallurgical Coatings and Thin Films 2004
April 22, 2004
San Diego, California, USA

Generalized Thermal Barrier Coating Failure Modes

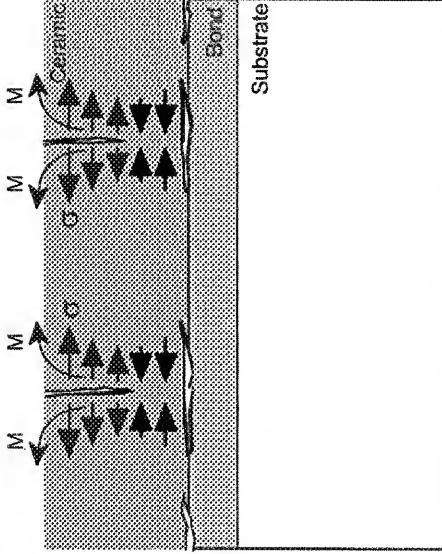
- Crack propagation is a critical issue especially under surface heat flux, thermal gradient cyclic loading



(a) High Heat Flux and
Low Interface
Temperature



(b) Low Heat Flux and
High Interface
Temperature



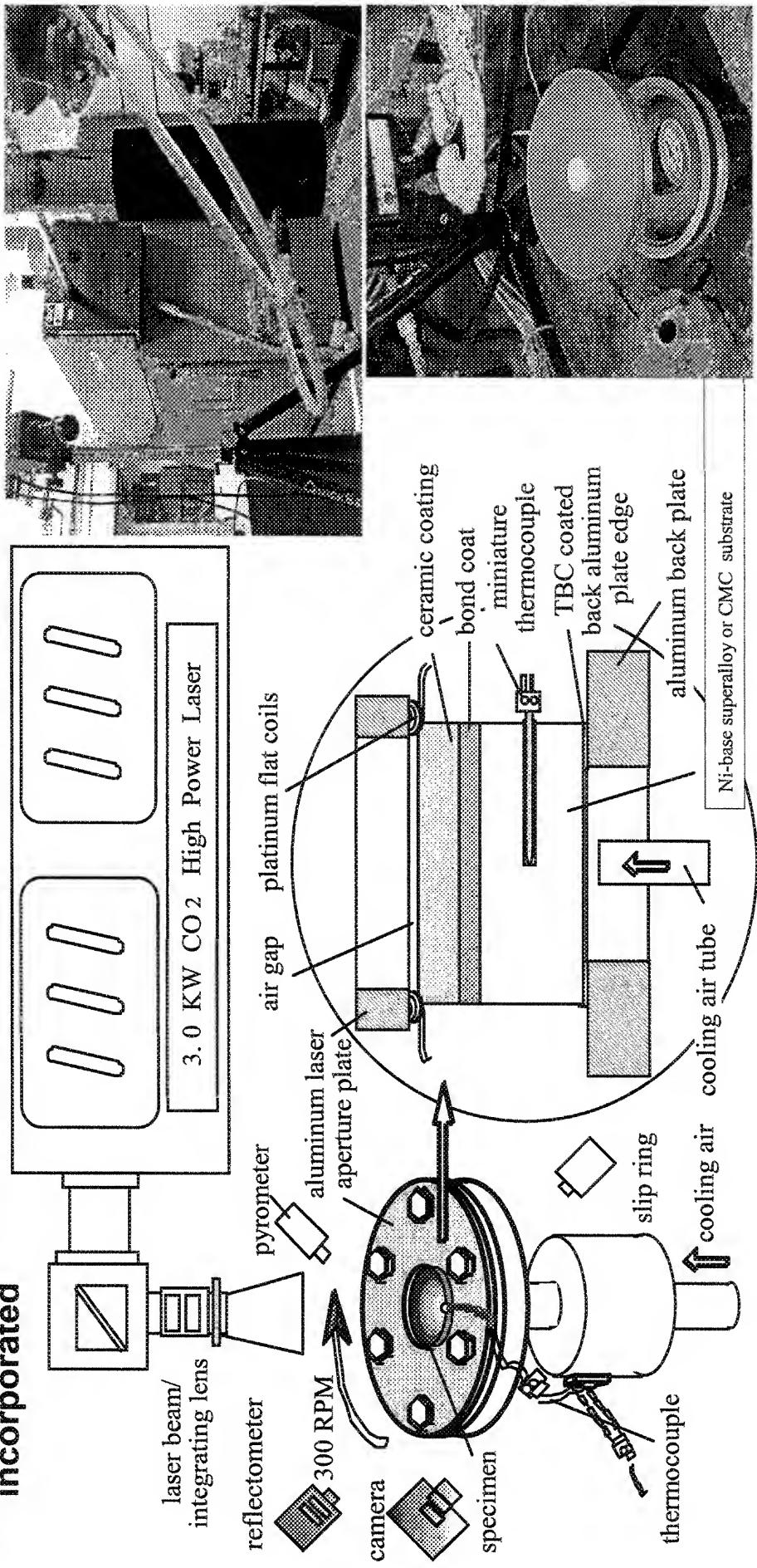
(c) Medium Heat Flux
and Interface
Temperature

Objectives

- Investigate the coating crack propagation under realistic high temperature and thermal gradient cyclic loading
- Laser heat flux testing for advanced thermal barrier coatings development

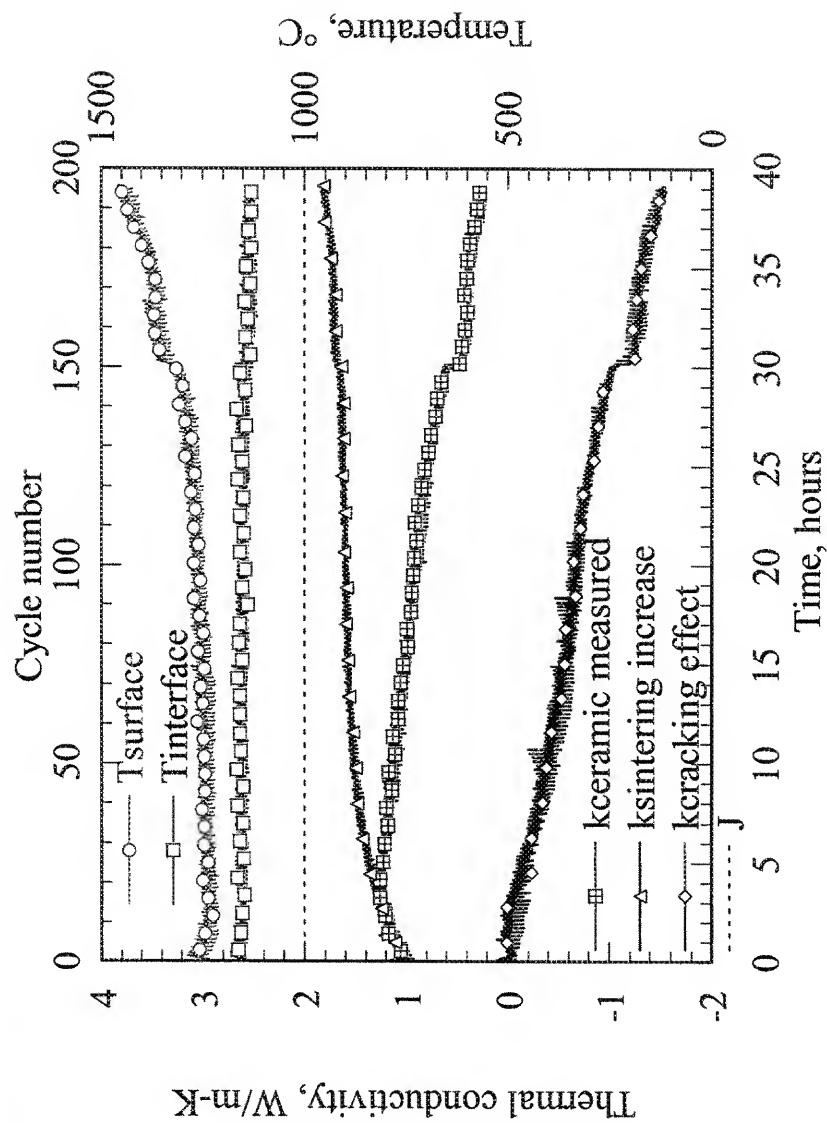
Laser Heat Flux Technique used for the Coating Real-time Crack Propagation Study

- A uniform laser (wavelength $10.6 \mu\text{m}$) power distribution achieved using integrating lens combined with lens/specimen rotation
- The ceramic surface and substrate temperatures measured by 8 micron and two-color pyrometers and/or by an embedded miniature thermocouple
- Thermal conductivity measured at 5 second intervals in real time and thermal cycling incorporated

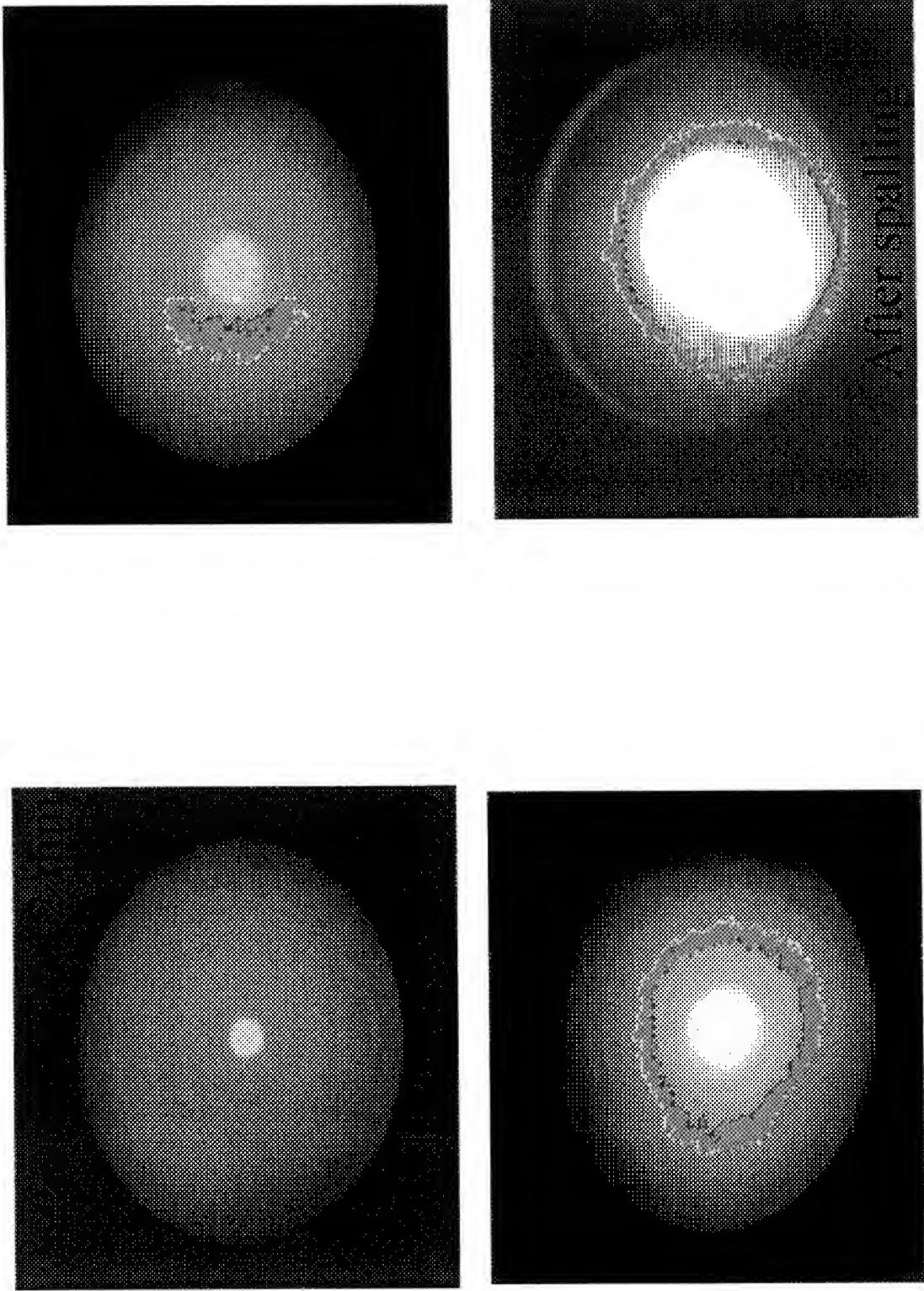


Temperature Response and Thermal Conductivity Changes of ZrO_2 -8wt% Y_2O_3 as a Function of Cycle Number

- Coating tested under 10 min heating and 2 min cooling laser thermal cycling condition
- Surface temperature increases and the metal backside temperature decreases as the delamination crack is initiated and propagated
- Coating conductivity initially increases due to coating sintering and then decreases due to crack propagation

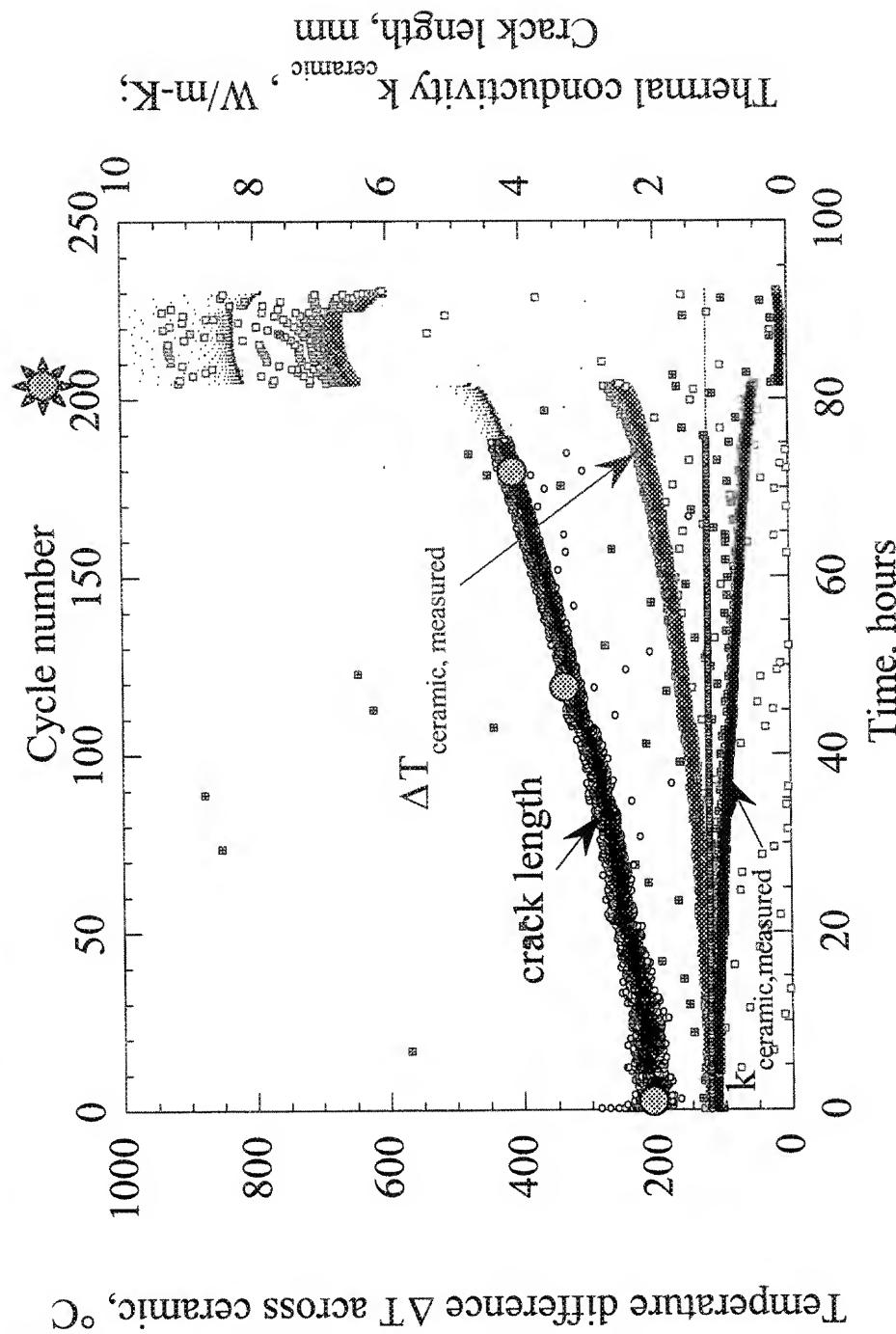


Crack Propagation of ZrO_2 -8wt% Y_2O_3 System — 0.2 mm Thick TBC Specimen with a 2-mm hole in the Substrate

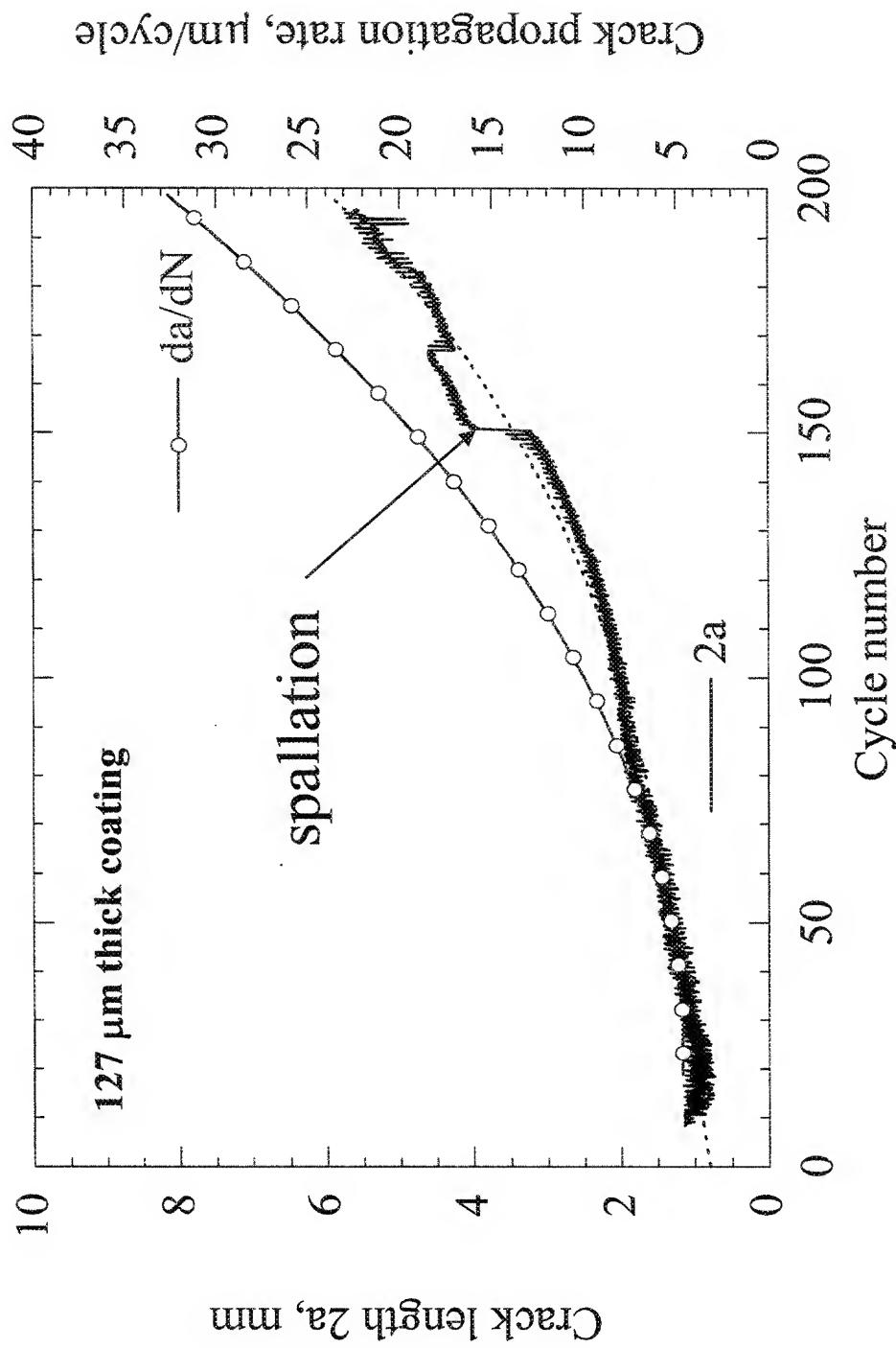


The Laser Thermal Fatigue Test Results of A 0.2 mm Thick $\text{ZrO}_2\text{-}8\text{wt\%Y}_2\text{O}_3$ with a 2 mm hole in the Substrate

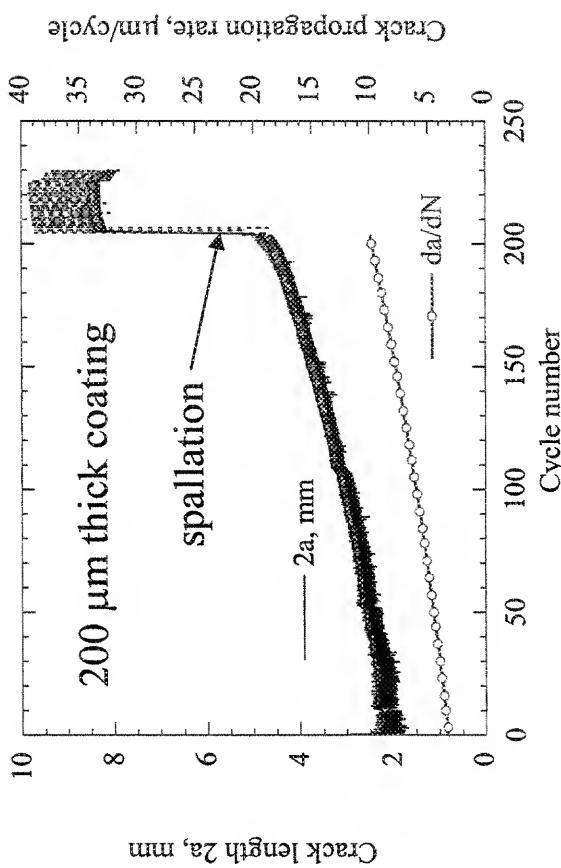
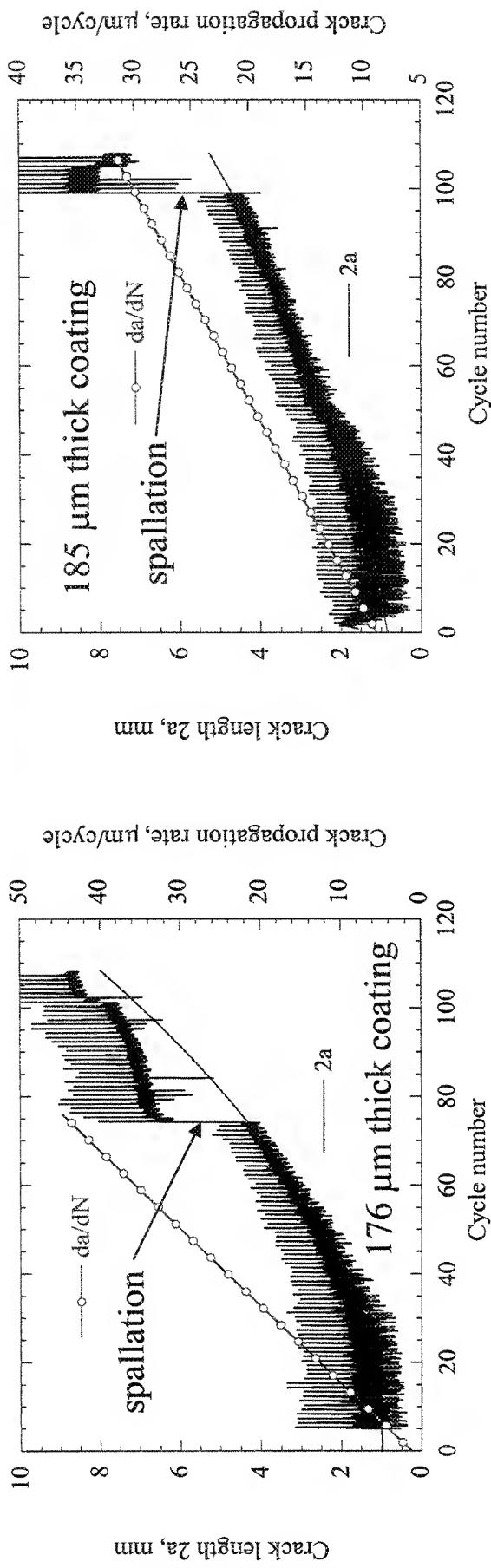
- Specimen exposed to 20 min heating and 4 min cooling laser cycling
- A close relationship between the coating conductivity and delamination crack length demonstrated



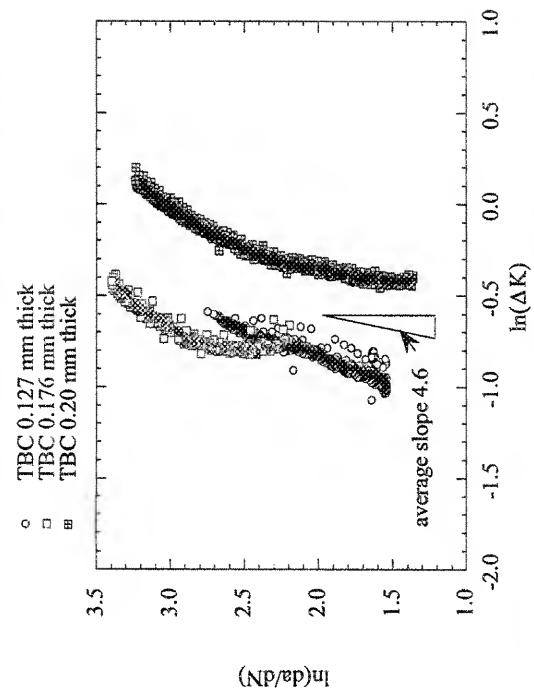
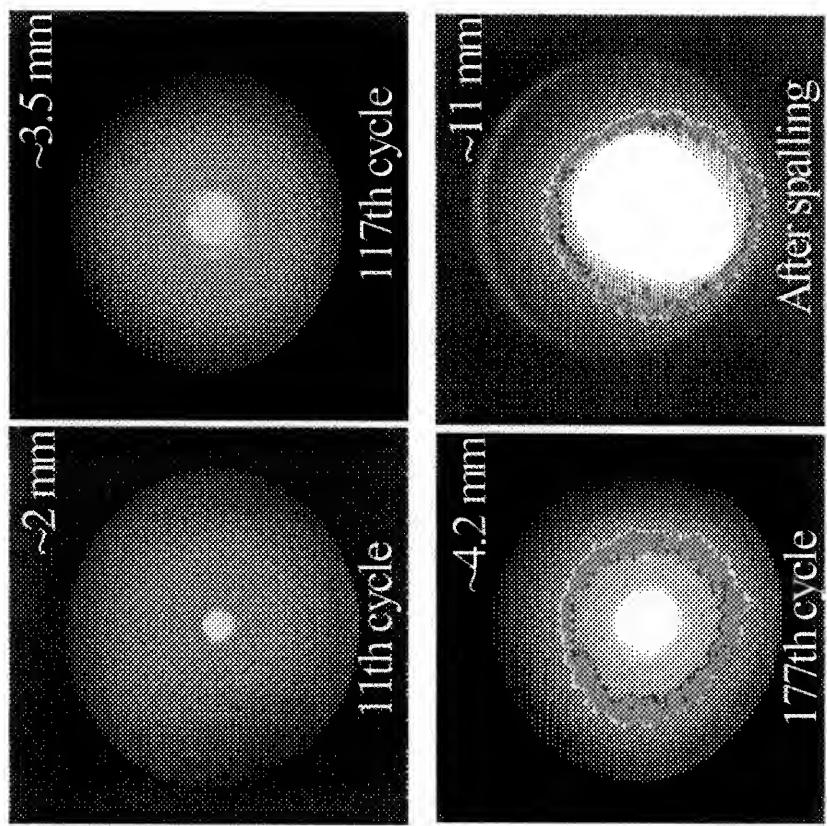
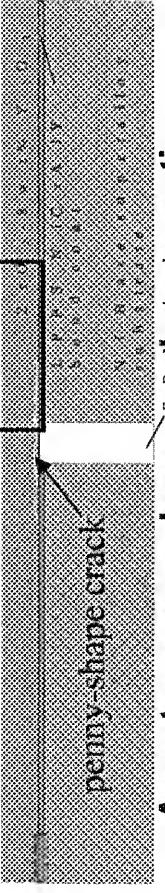
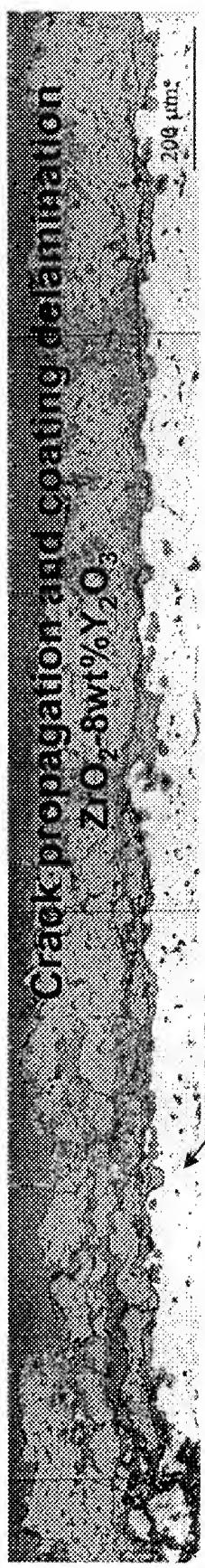
Crack Length and the Crack Propagation Rates of ZrO_2 - 8wt% Y_2O_3 as a Function of Cycle Number



Crack Length and the Crack Propagation Rates of ZrO₂-8wt%Y₂O₃ as a Function of Cycle Number



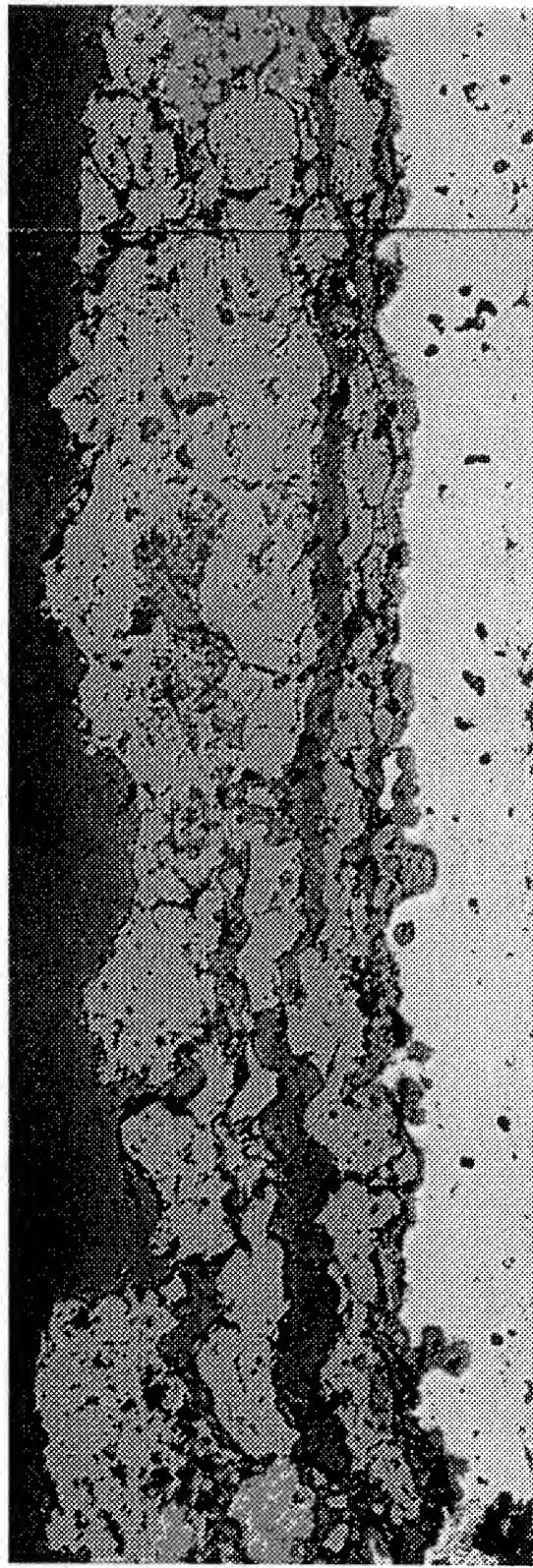
The Relationship between Crack Propagation Rate and Laser Thermal Stress Associated Stress Intensity Factor Amplitude



Crack propagation $d\alpha/dN$ -stress intensity amplitude ΔK plot for life prediction

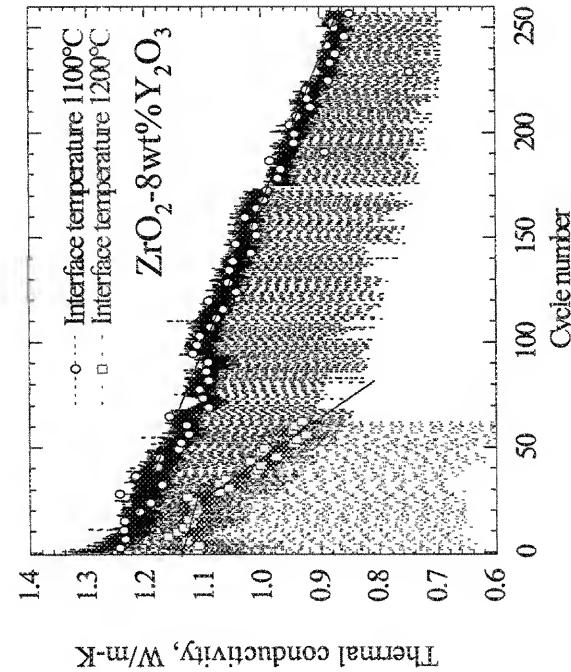
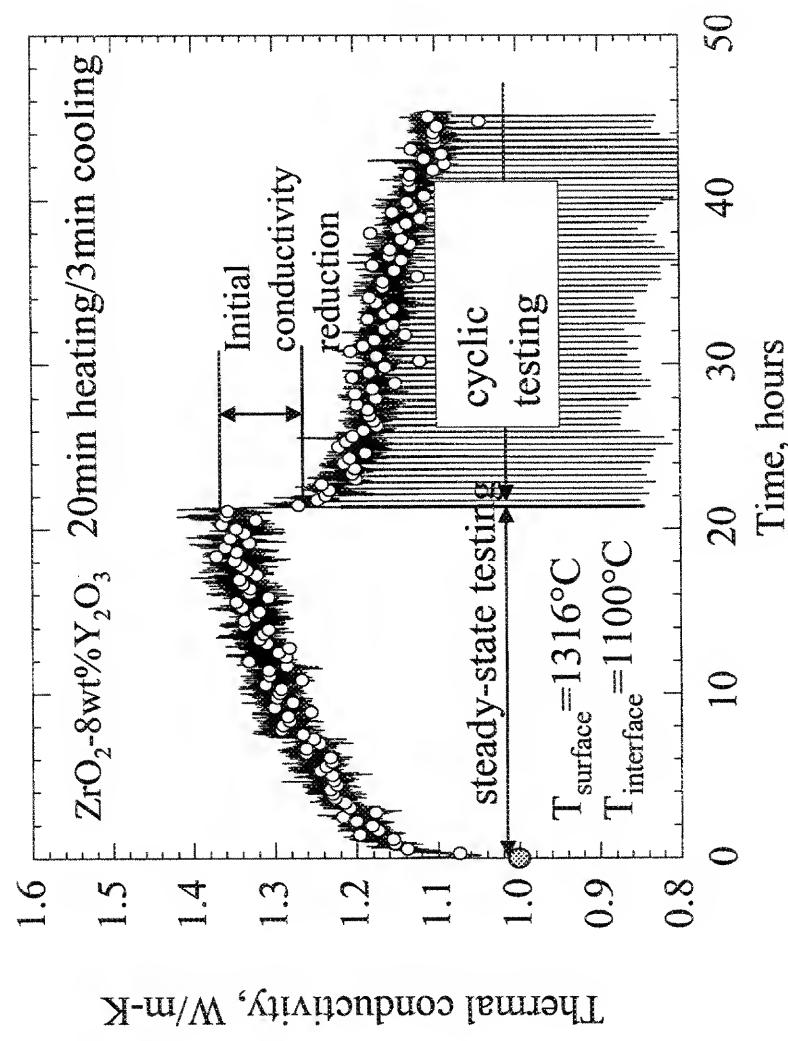
Micrograph of Laser Thermal Fatigue Tested TBC Showing Coating Delamination Crack Propagation

- Severe fatigue damages are observed near the early crack propagation wake surfaces
- Strong coating asperity/debris interactions and coating multiple delaminations
- The later crack paths show relatively smooth surfaces, which corresponds to the faster crack propagation regions under the increased crack propagation driving force



Laser Heat Flux Steady-State and Cyclic Testing of ZrO₂-8wt%Y₂O₃ Thermal Barrier Coatings

- Thermal conductivity monitoring used for coating durability evaluation
- ZrO₂-8wt%Y₂O₃ coating cyclic durability demonstrated an issue at 1316°C



Development of Advanced Defect Cluster Thermal Barrier Coatings

- Develop low conductivity and high stability thermal barrier coatings using oxide defect clustering approach
- Selected multi-component clustered oxide TBC systems investigated and reported - NASA UEET low k coating systems



- Real-time monitoring and evaluation of the coating thermal conductivity and sintering
- Furnace cyclic oxidation test for initial coating durability assessment
- Higher temperature heat-flux cyclic tests for temperature durability/capability test

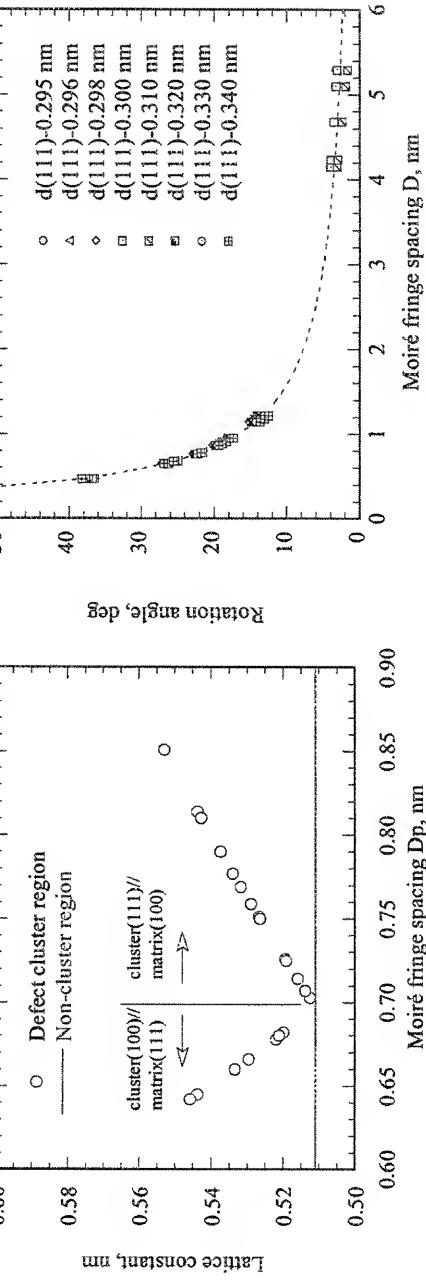
Defect Clusters Identified using High Resolution Transmission Electron Microscopy and EELS Analysis

- The 5 to 100 nm size defect clusters are believed to be responsible for the reduced thermal conductivity and improved stability



Plasma-sprayed ZrO_2 -
13.5mol%($\text{Y}, \text{Nd}, \text{Yb}$) $_2\text{O}_3$

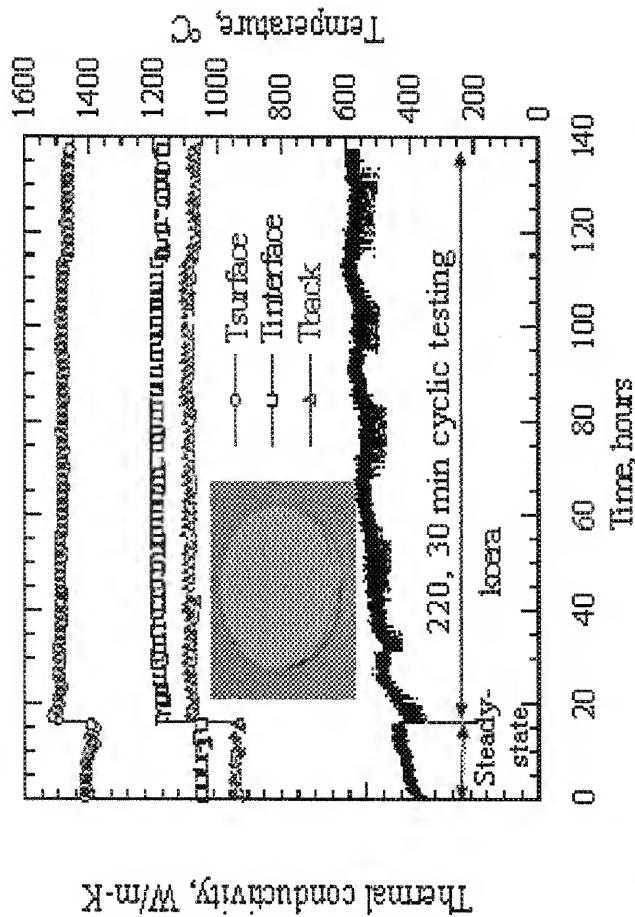
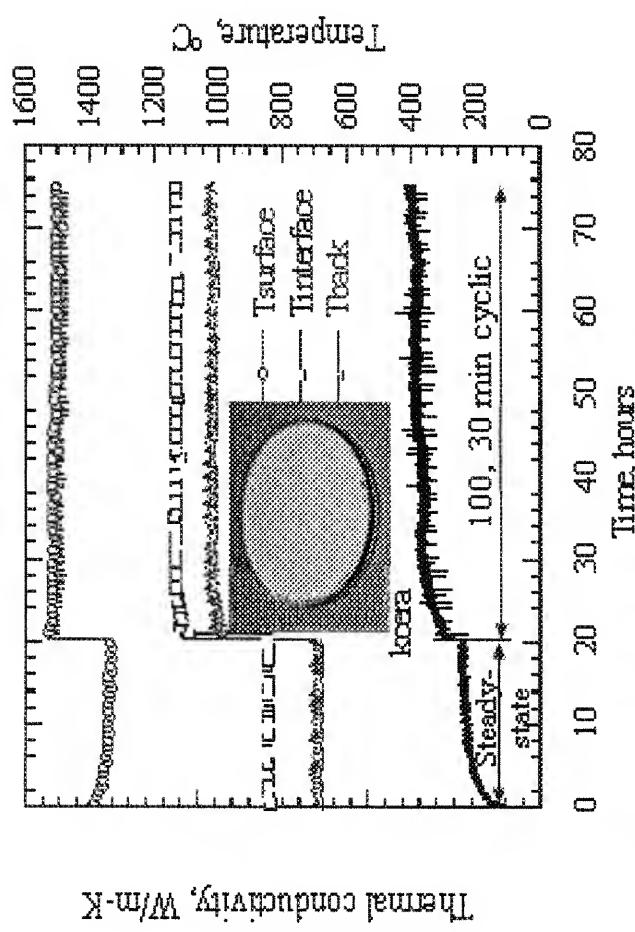
EELS elemental maps of EB-PVD ZrO_2 -
14mol%($\text{Y}, \text{Gd}, \text{Yb}$) $_2\text{O}_3$



Lattice constant and
rotation angle of the
defective and dopant
segregated lattice
characterized

Sintering and Cyclic Durability of Advanced TBCS

- The advanced defect cluster coatings demonstrated very high temperature cyclic durability



(a) The NASA 'l' low k coatings

- 20h Sintering:
 - $T_{\text{surface}} = 1360^{\circ}\text{C}$ (2500°F), $T_{\text{interface}} = 850^{\circ}\text{C}$ (1562°F)
 - 100, 30 min cyclic testing:
 - $T_{\text{surface}} = 1535^{\circ}\text{C}$ (2795°F), $T_{\text{interface}} = 1135^{\circ}\text{C}$ (2075°F)

(b) NASA low k/advanced interlayer coating

- 20h Sintering:
 - $T_{\text{surface}} = 1400^{\circ}\text{C}$ (2552°F), $T_{\text{interface}} = 1050^{\circ}\text{C}$ (1922°F)
 - 220, 30 min cyclic testing:
 - $T_{\text{surface}} = 1510^{\circ}\text{C}$ (2750°F), $T_{\text{interface}} = 1180^{\circ}\text{C}$ (2156°F)

Conclusions

- A laser thermal fatigue approach has been established to study the delamination crack propagation of thermal barrier coatings.
- Real-time monitoring of coating thermal conductivity demonstrated as an effective technique to assess coating performance under simulated engine conditions.
- For the ZrO_2 -8wt% Y_2O_3 coating specimens tested, the initial average crack propagation rate was in the range of 3-8 $\mu\text{m}/\text{cycle}$. The crack propagation rates increased to 30-40 $\mu\text{m}/\text{cycle}$ at the later stage of the tests. The accelerated crack growth is attributed to the increased driving force for the crack propagation under the laser heat flux cyclic test conditions.
- The multi-component advanced TBCs demonstrated significantly improved long-term high temperature stability and cyclic durability at very high temperatures required for advanced turbine airfoil and combustor applications.